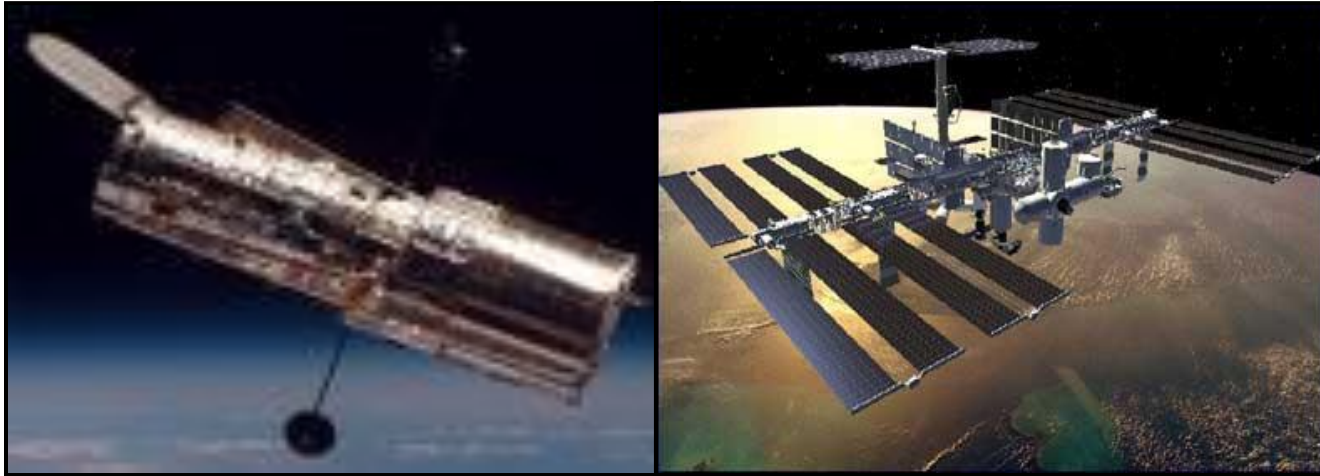


SABER

ASTRONAUTICS

Next Generation of Space Control

Where we come from



Based in Sydney, Boulder, CO USA

Prior work flight software on Hubble and ISS

Senior operators from USSTRATCOM

PhD graduates from world class robotics labs

*+ a mature internship program
150 applicants per year*



The Problem: Bespoke Software

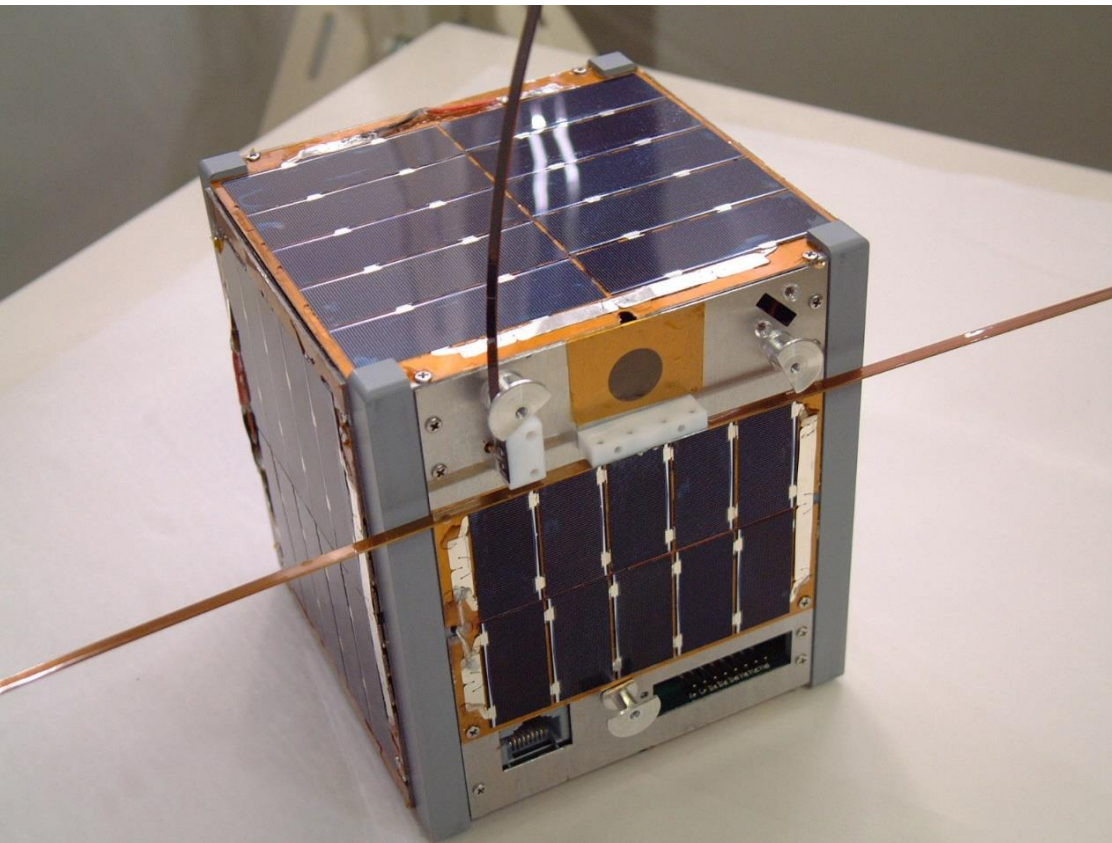
Both in space and on the ground



Problem: Diagnostics

Because every minute of satellite downtime is millions in lost revenue

A single CubeSat has 100 health sensors



Effects:

- **Single Event Upsets**
- **False Positives**
- **Diagnostic Challenges**

What is the root cause of damage?

$2^{n(n-1)/2}$ causes

2^{4950} possibilities!

(sorry for the math folks. Lets just say "infinite" and move on)



The process is like this:

"You have to knead your own bread to make a sandwich"



Satellite Design
Mission Planning
Dish commands
Satellite commands
Error tracking
Alert Systems
Databasing
Diagnostics
Ops software



A satellite mission



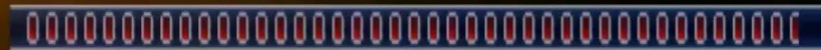
Solution - PIGI

www.fraps.com

P · I · G · I

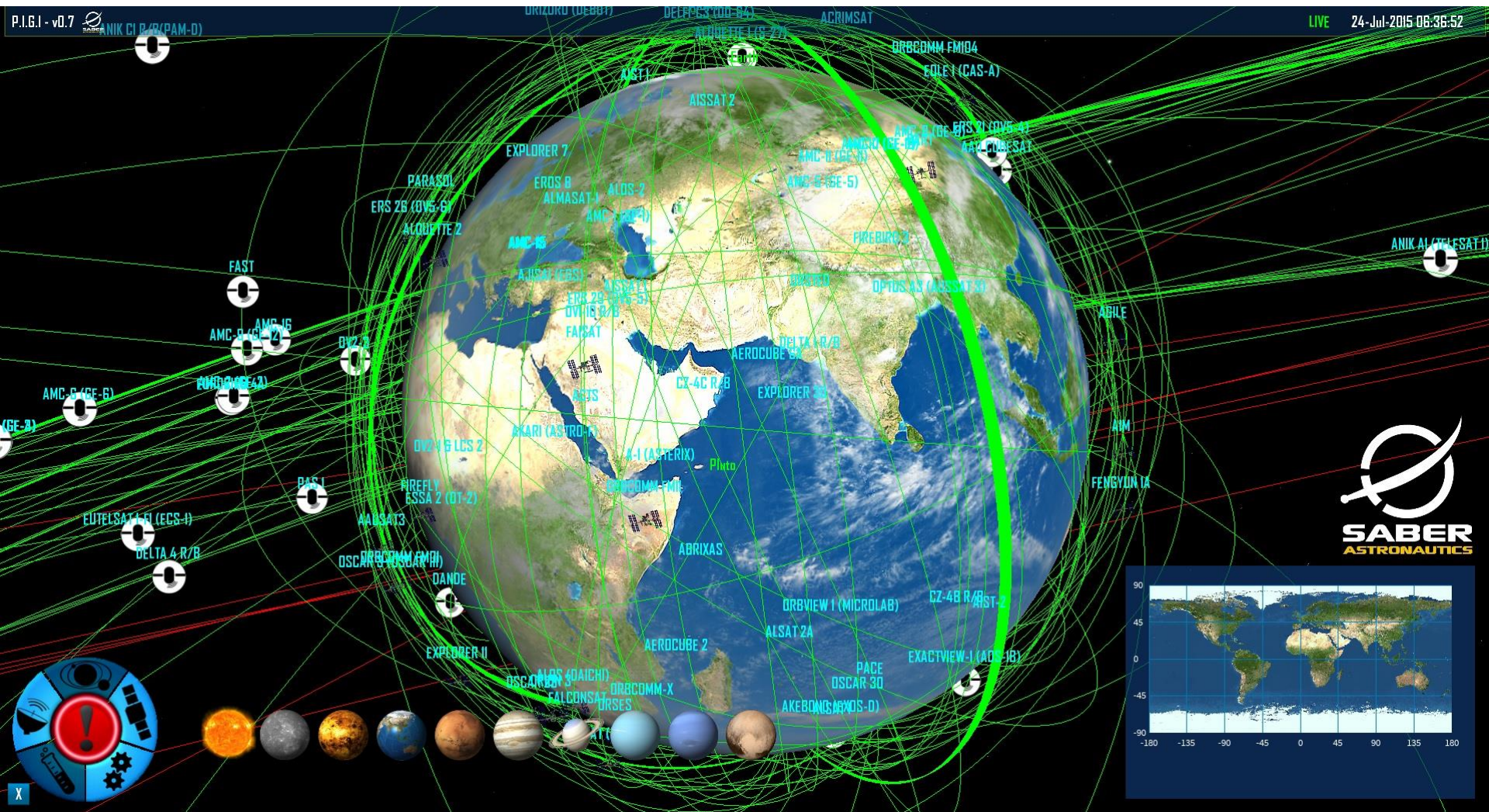
PREDICTIVE INTERACTIVE GROUND STATION INTERFACE

LOADING HIPPARCOS...



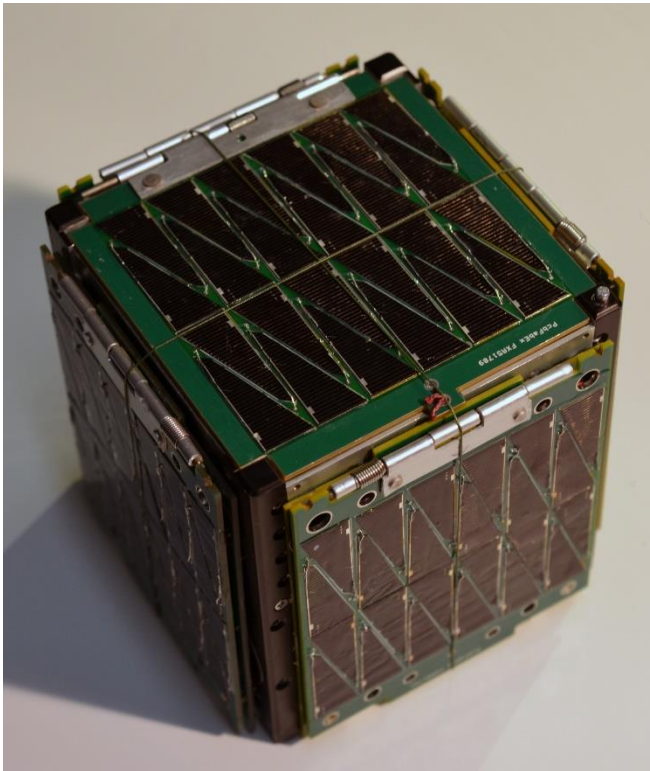
Now multiply that by 100

130 satellite constellation viewed in PIGI



Testimonial - SkyCube

"Level 2 Ops" from 2014-03-12 and 2014-07-09



Integrated software, graphics benefit

Direct observation and situational awareness

"You had twice as many connections as both active MC3 sites combined. "



MC3 is the US Navy Small Satellite network

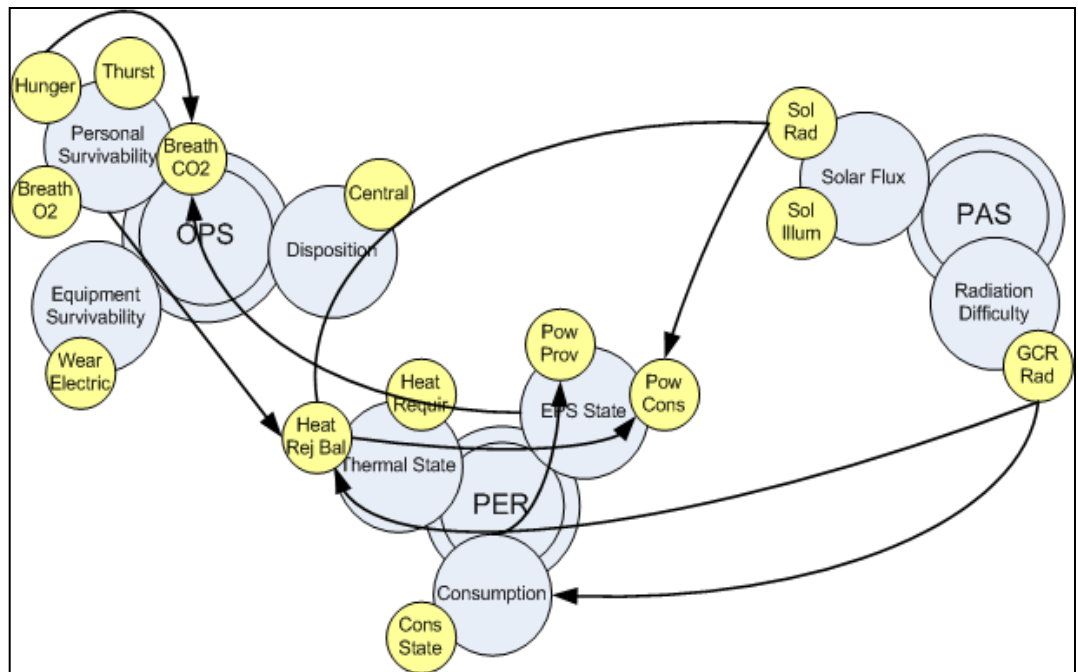


System Map

Model metric interactions

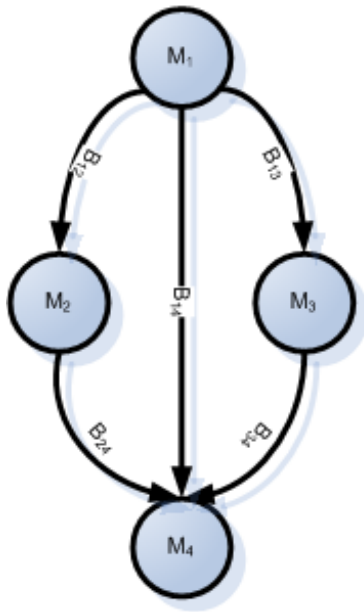
Options

- ☐ MLR
- ☐ Neural Networks
- ☐ Genetic Algorithms
- ☐ Bayesian Networks (DBN)



Bayesian Networks (BN)

- Represent each metric as a Gaussian
- If you know the connections, find parameters

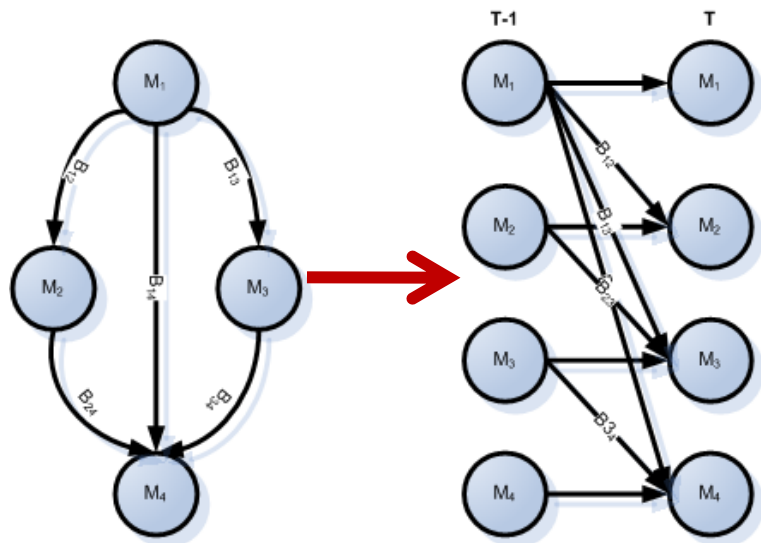


$$P(M_1, \dots, M_n) = \prod_i P(M_i | Pa(M_i))$$

$$\theta_{ML} = \arg \max_{\theta} \log(Z_L | \theta)$$

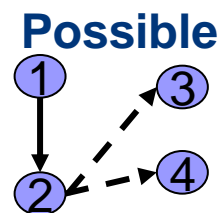
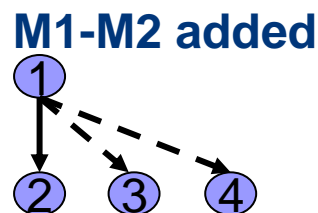
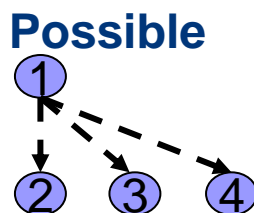
Dynamic Bayesian Networks

- 1st Order Markovian assumption

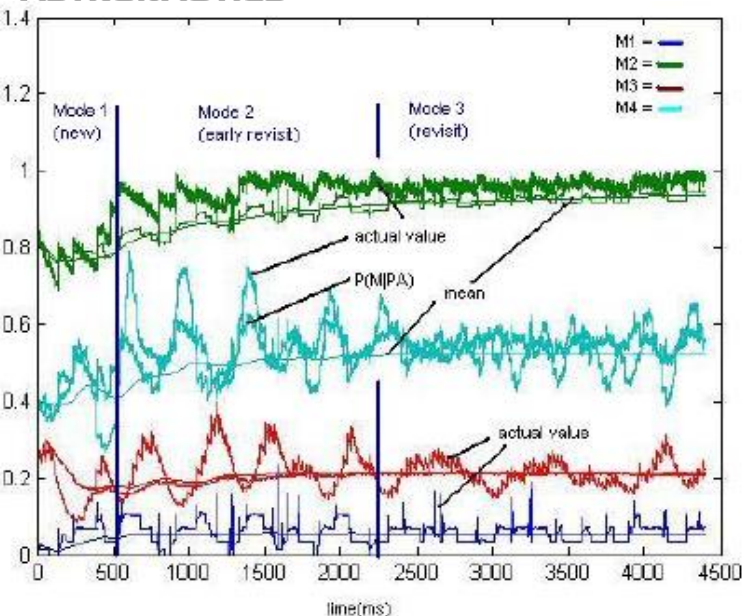


$$P(Z_{1:T}) = \prod_{t=1}^T \prod_{i=1}^N P(Z_t^i | Pa(Z_t^i))$$

- Unknown structure, full observability
- Search using a scoring function



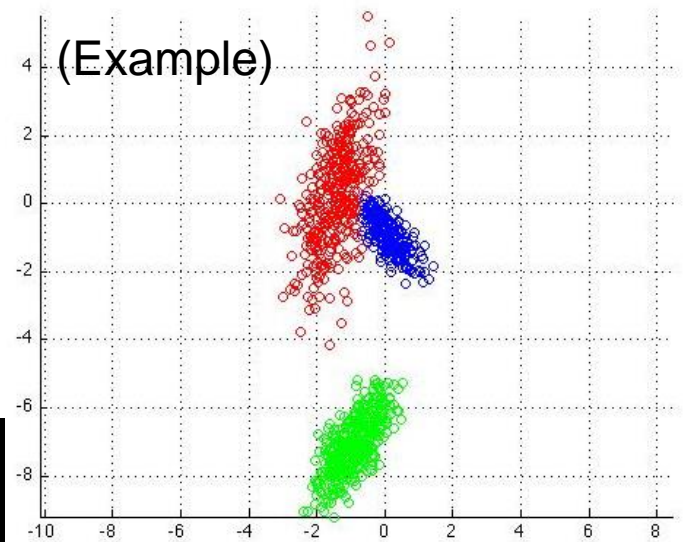
Switching States



Identify a change in mode

- ☐ Multiple (unknown) hidden states
- ☐ Single hidden state (EM)
- ☐ **GMM (EM)**

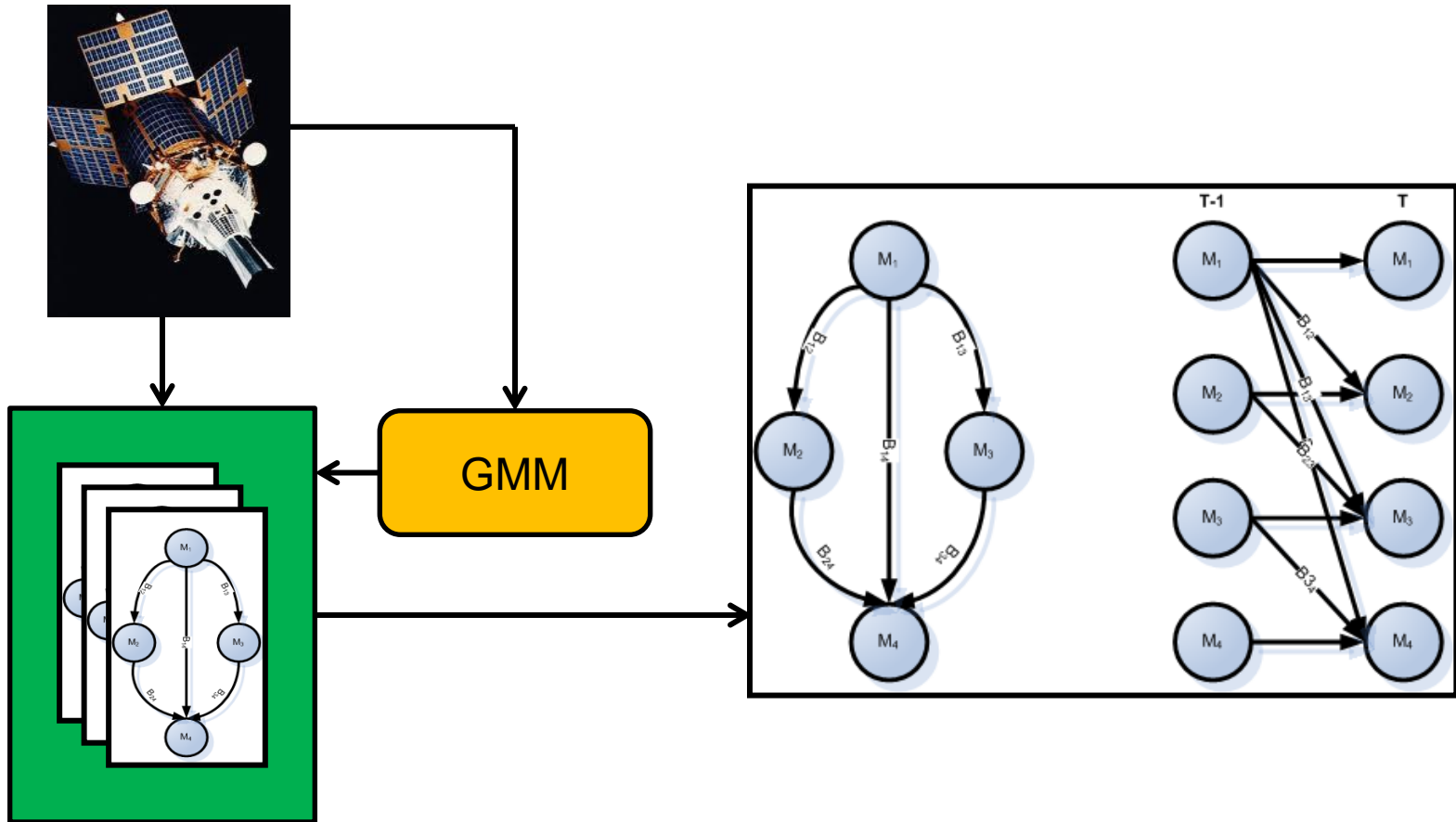
$$P(M_i | \mu_1, \dots, \mu_k, s_1, \dots, s_k, \pi_1, \dots, \pi_k) = \sum_{j=1}^k \pi_j N(\mu_j, s_j^{-1})$$



- ☐ Each mixture with its own DBN



At Run Time



NASA ACE Demo

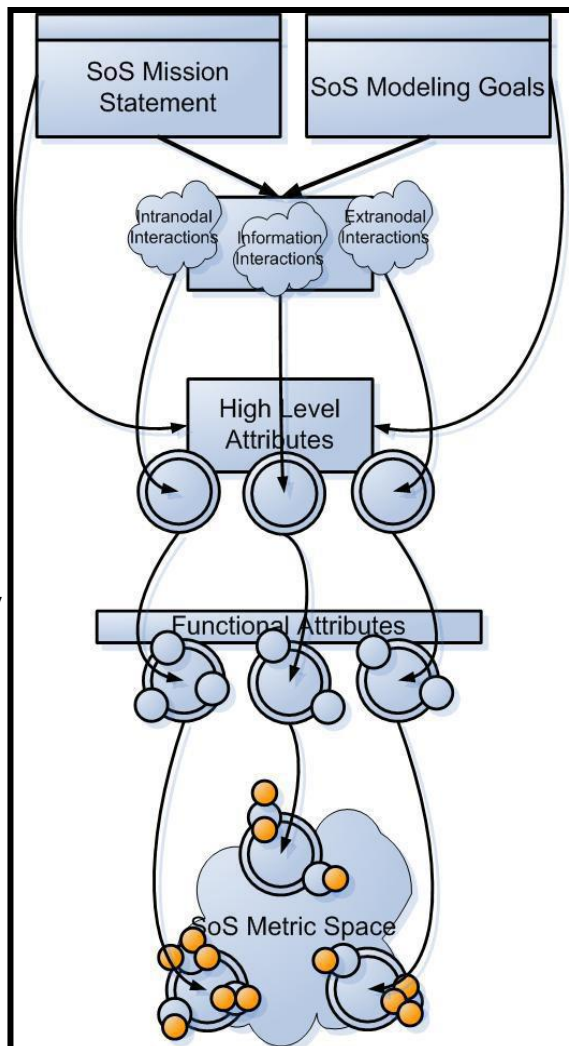


NASA GSFC Support

NASA ACE H/W Telemetry

1200 Telemetry Items

SEC Data



120 words to model
22 Health Metrics
3 SPWX Metrics

Experiments

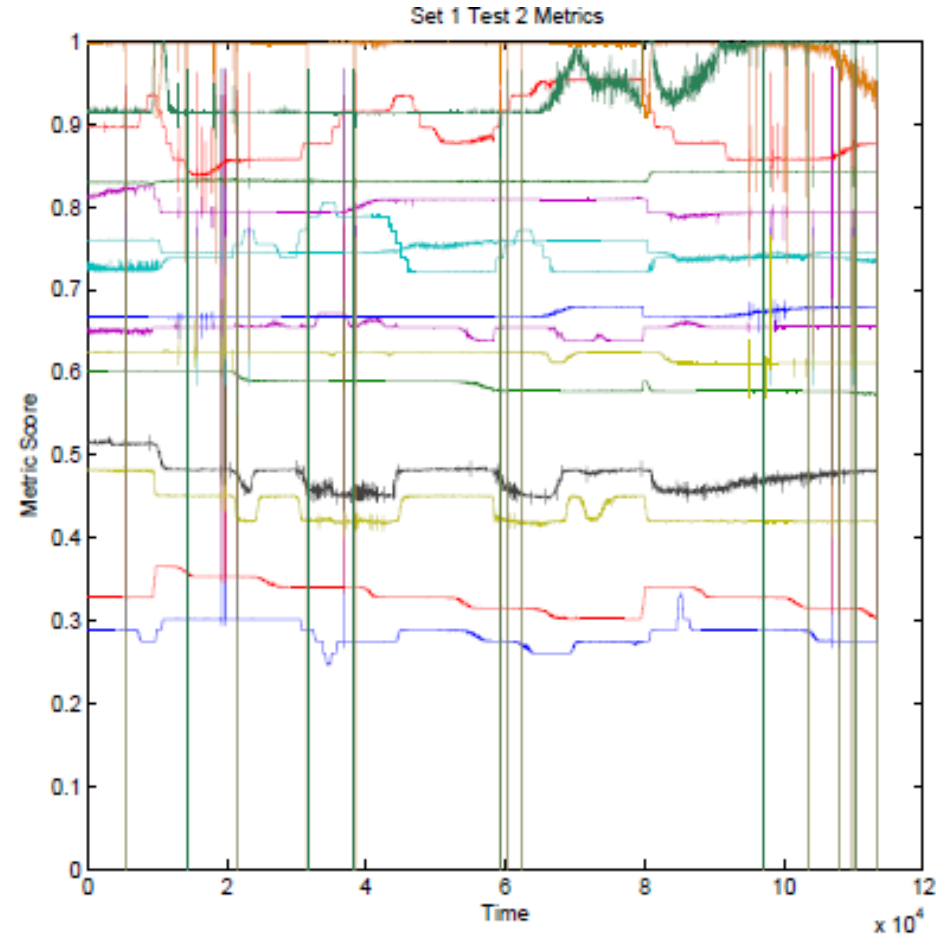
Training Data – 2003 Solar Flare

Long Distance Nominal (2009)

Long Distance Hazard (2011)

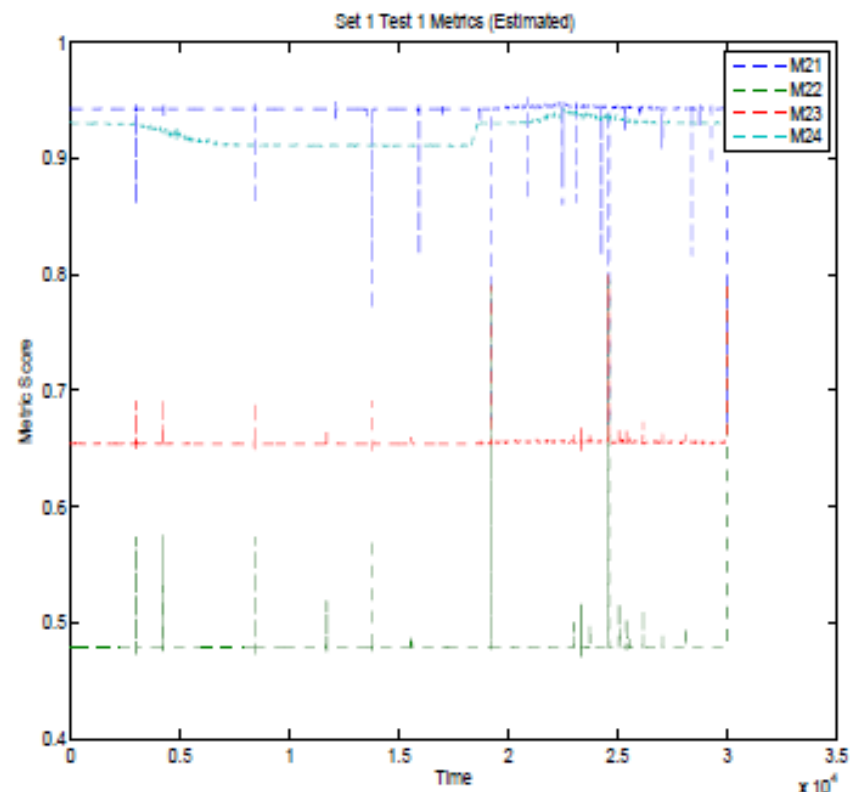
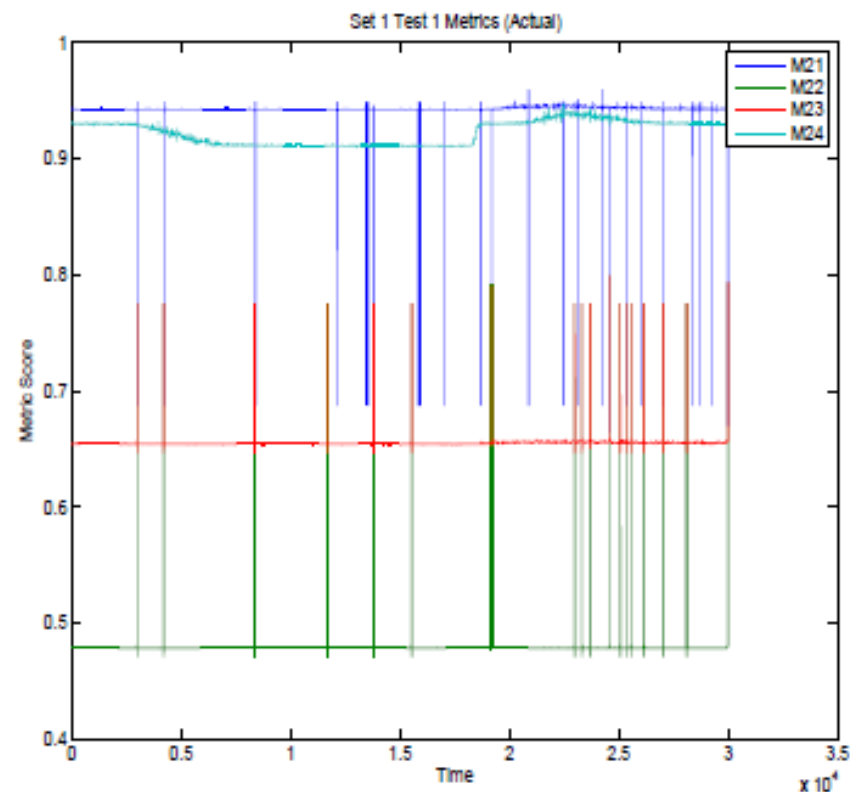
2-weeks of training data

8-years of experiment data



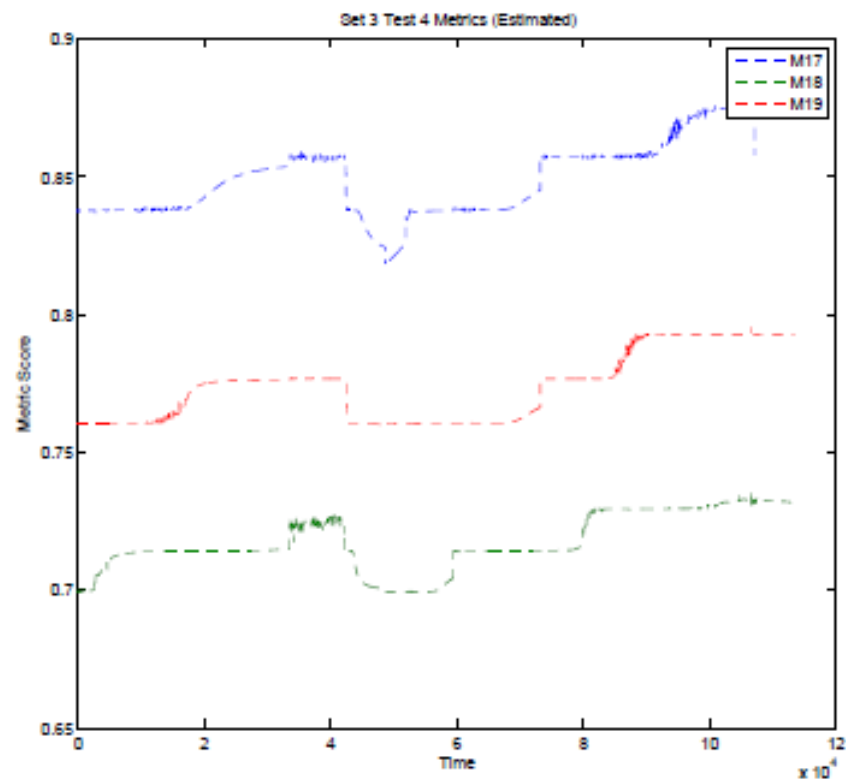
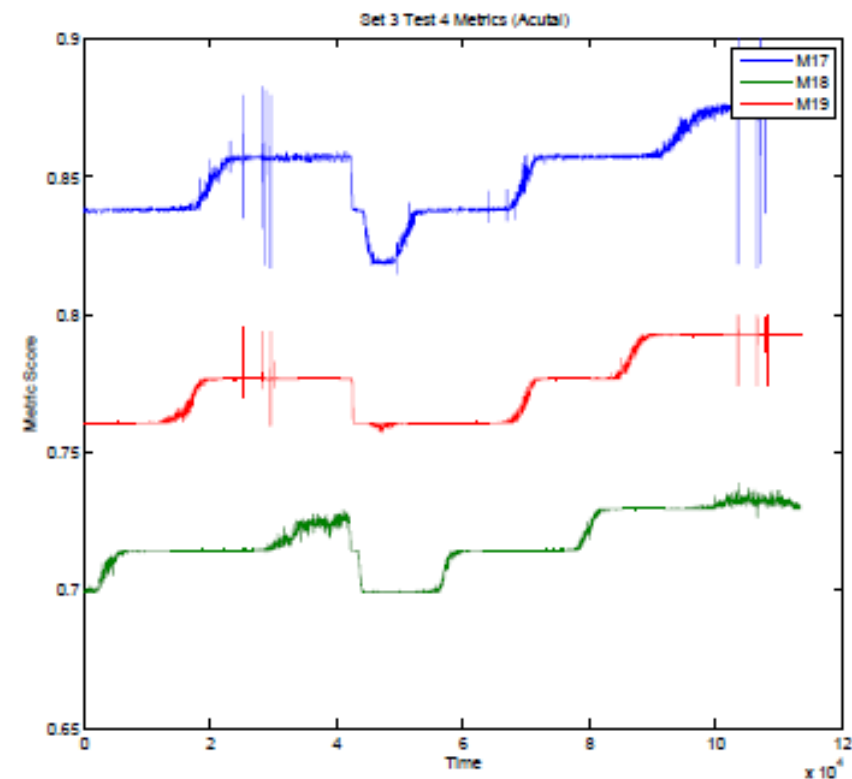
Results – Nominal tests 2009

Signs of a general solution, but more work to do in SpWx metrics



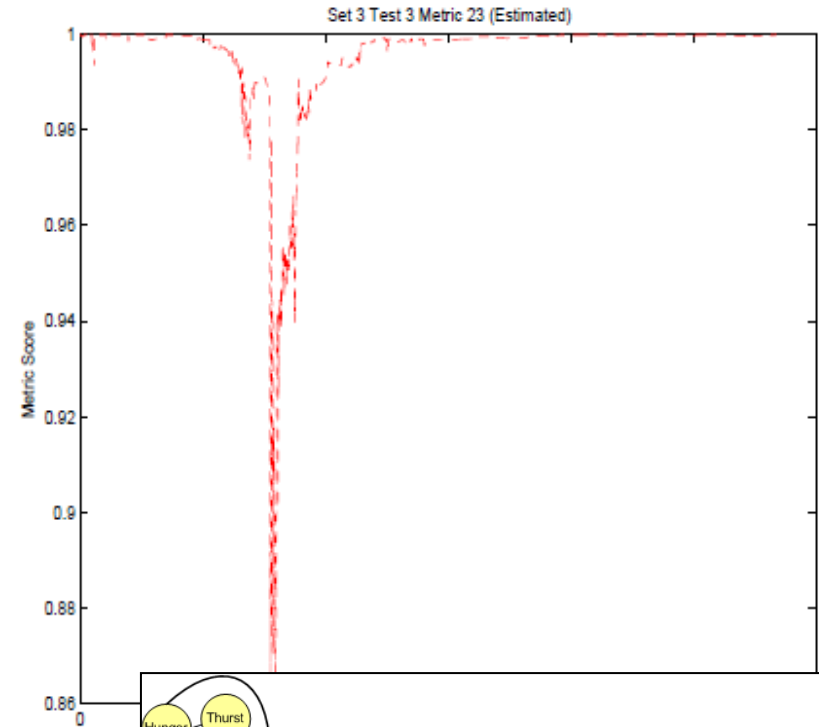
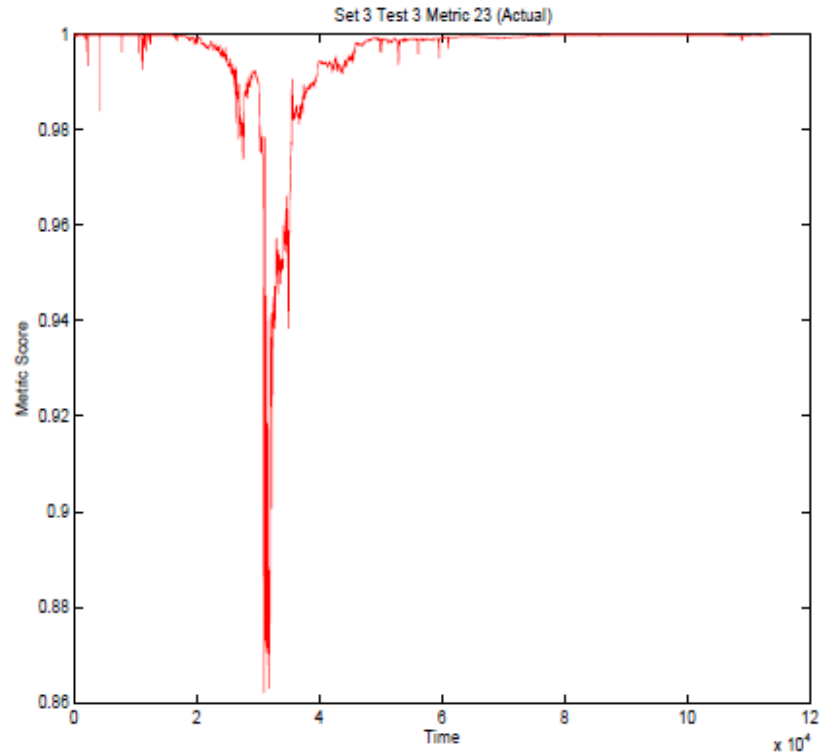
Results – Hazard tests 2011

Metrics 17, 18, 19 – Payload Heat Capacity



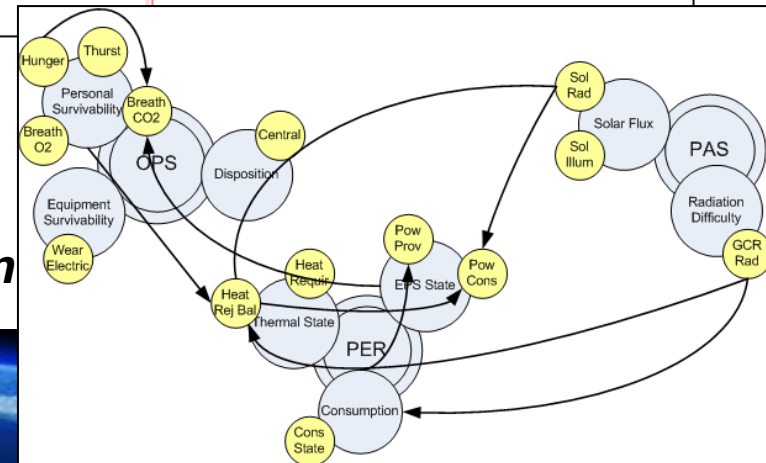
Results - Great

Metric 23 – Electron Hazard

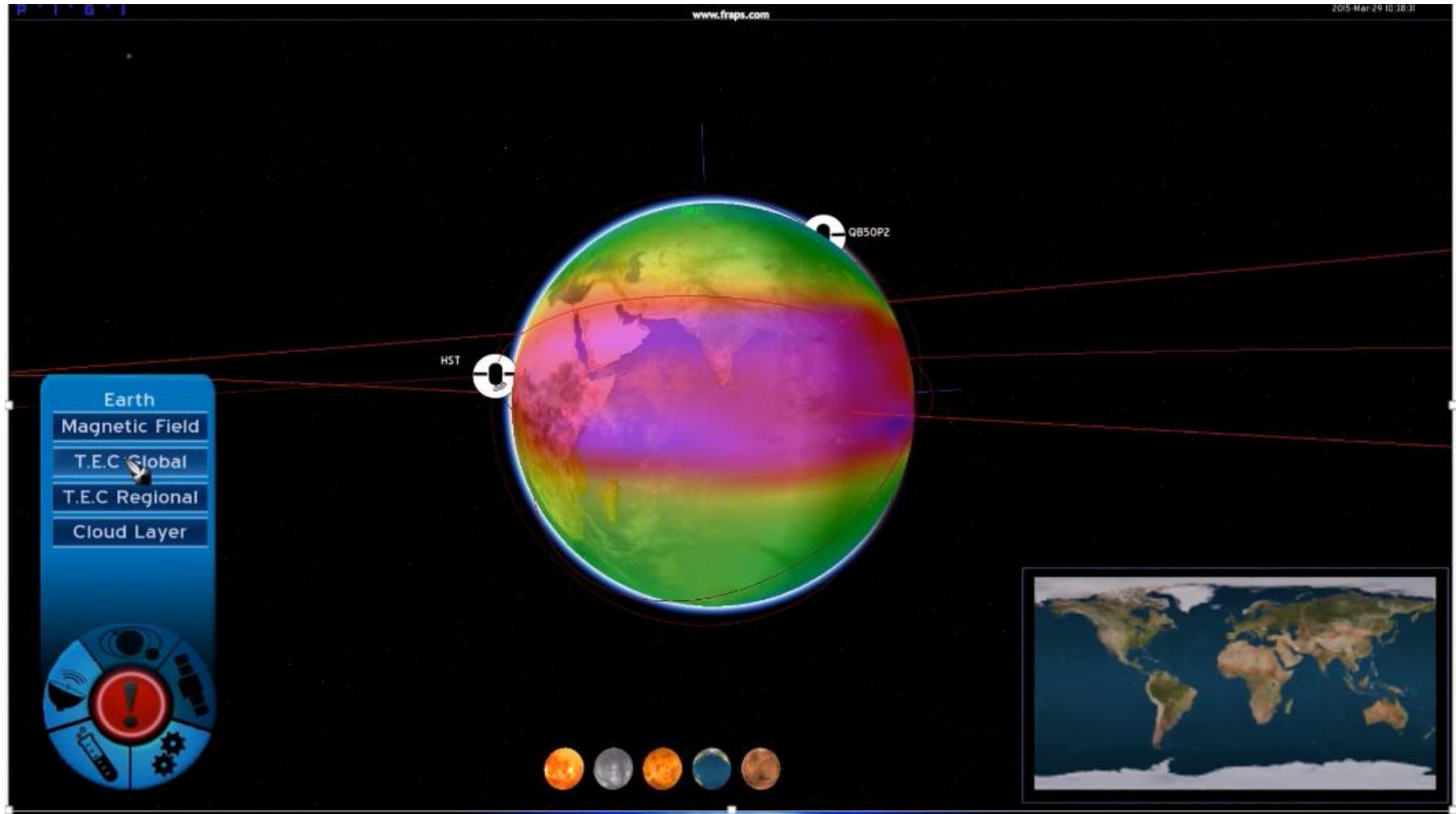


Time Invariant

Some aspects of a general solution



Real time services...

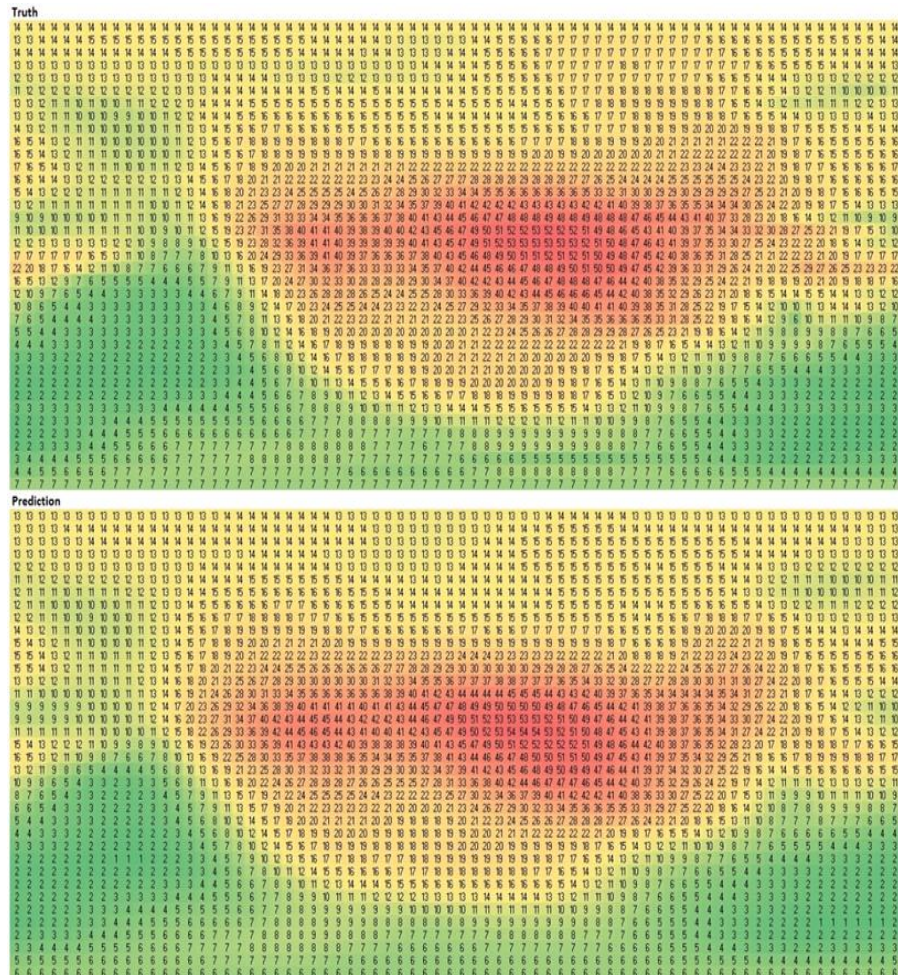


Space Wx prediction

Space Weather Prediction Service

- We can now predict Space Weather 1-3 hours ahead (92%-94%)
- This allows for spacecraft health prediction, preventative safing, safety operations, etc.

Services coming, stay tuned



Application: Morpheus

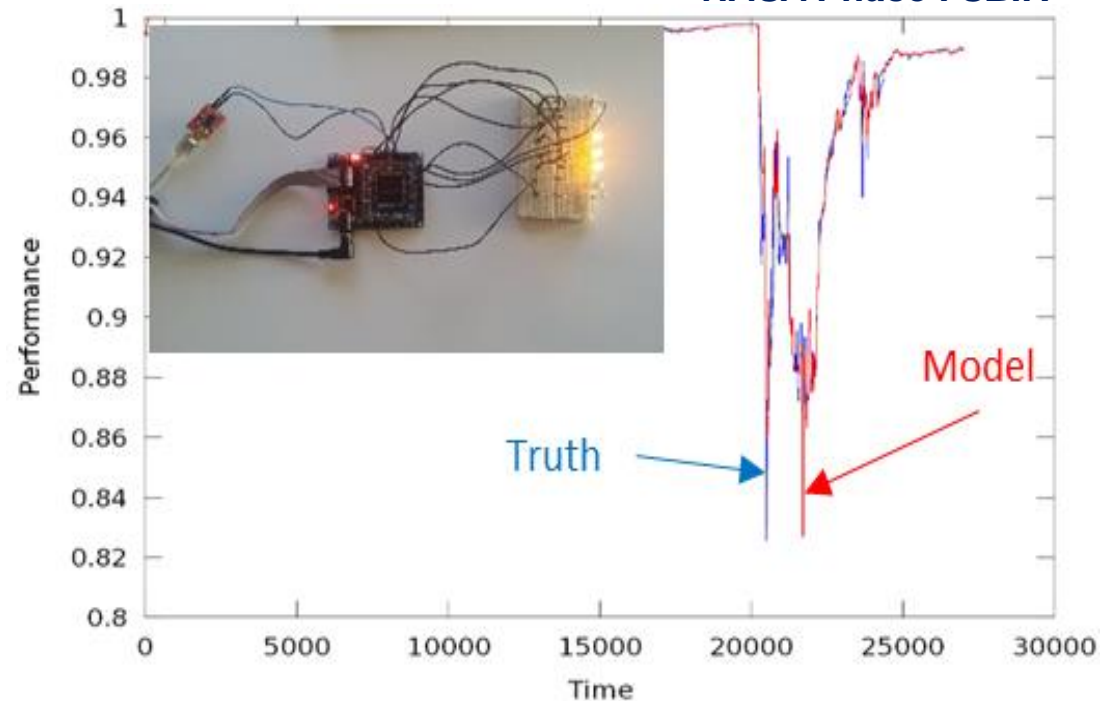
On-orbit, morphable avionics

Reacts to heat, cold, radiation

Common interface to PIGI

CASIS Grant \$200k

NASA Phase-I SBIR



Put it in space, show value, then bundle like hell



Application: Telescopes

Research question: can System Maps improve telescope performance?

1. Systems of Systems Analysis

What are the right metrics?

2. Prototyping on ground telescopes

Vibration

Thermal

Operational

Atmospheric

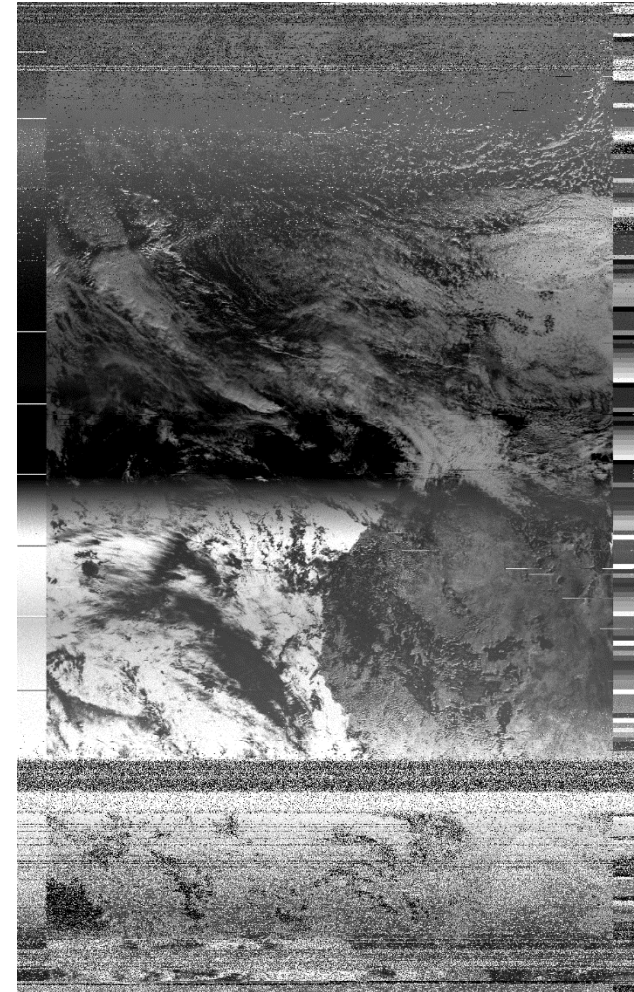
3. Environ chamber testing

Radiation

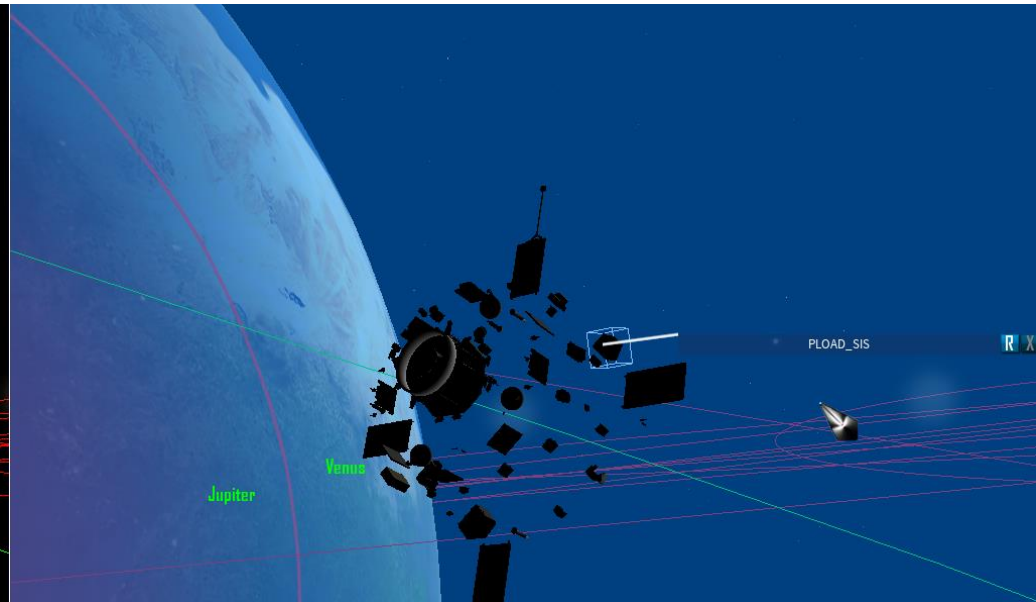
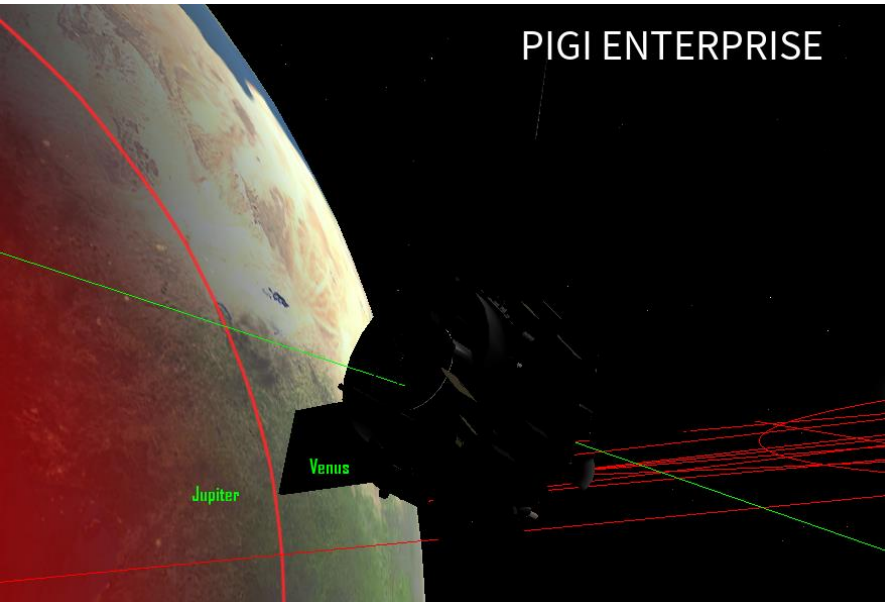
Thermal cycling

4. Flight

Telemetric, Ionospheric



Conclusion



Achieved 97-99% accuracy in systems observation.

SEUs identification accuracy 71-85%

Integrated fleet services